

**Kurdistan Regional Government-Iraq**

**Kurdistan Engineers union**

**Report Title**

**Steel Truss Design.**

**By**

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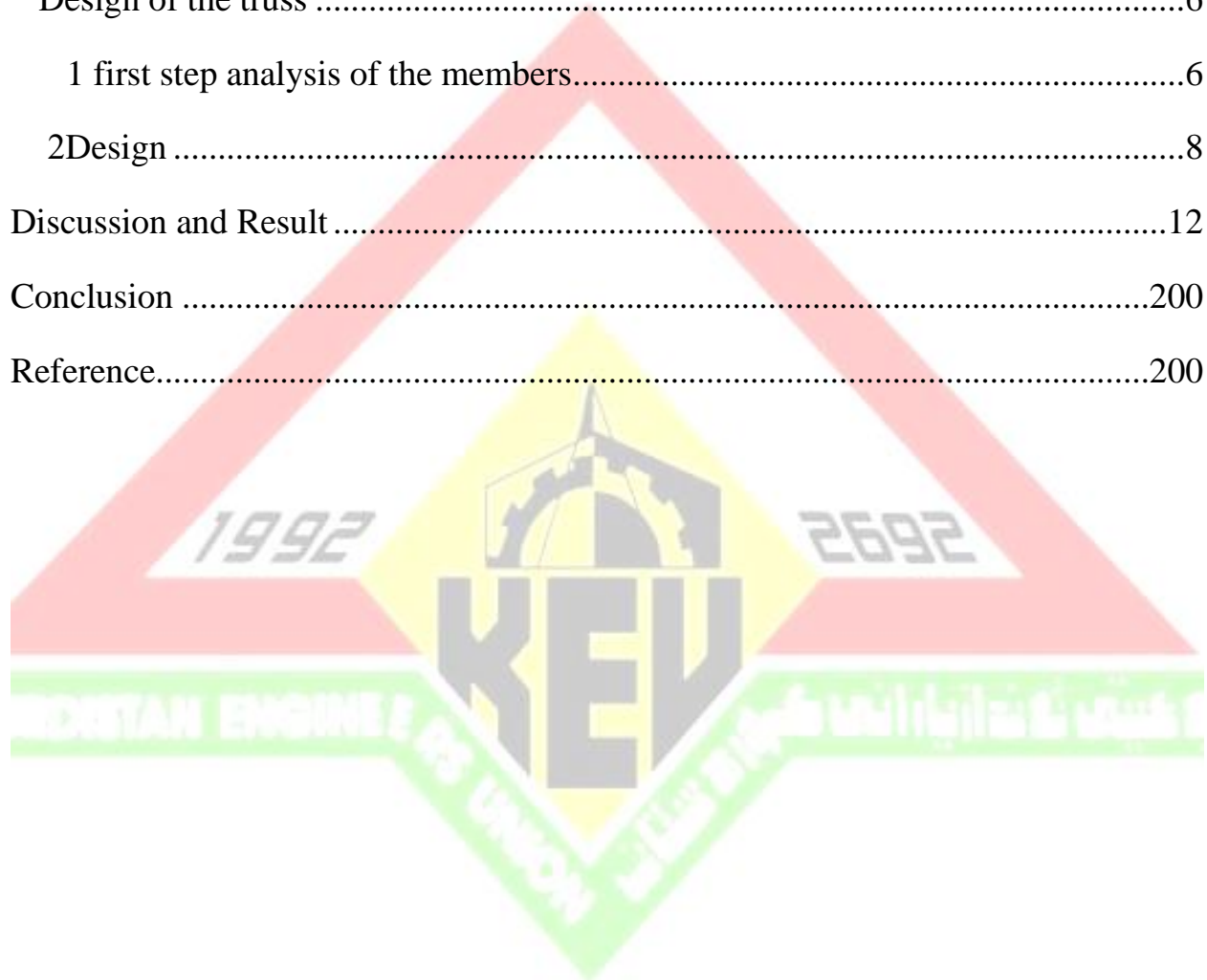
## Abstract

This study examines the design principles, materials, construction methods, and applications of steel trusses in modern architecture. It emphasizes the value of load distribution, material selection, and structural analysis while demonstrating how steel trusses provide dependable and efficient solutions for large-span constructions. The research provides an overview of the design processes, including everything from basic inspiration to detailed analysis and optimization using state-of-the-art software and analytical methodologies. Steel trusses are beneficial and advantageous in a variety of architectural scenarios, as demonstrated by case studies.



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# 1 Introduction

Steel trusses are an essential part of civil engineering and architectural projects because of their great load-bearing capability and ability to span huge distances with little material consumption. These structures are perfect for applications like roofs, towers, and bridges because they are made of linked triangle components that distribute stresses efficiently. The purpose of this paper is to present a thorough review of steel truss design and analysis, with an emphasis on important concepts, materials, techniques, and practical applications.



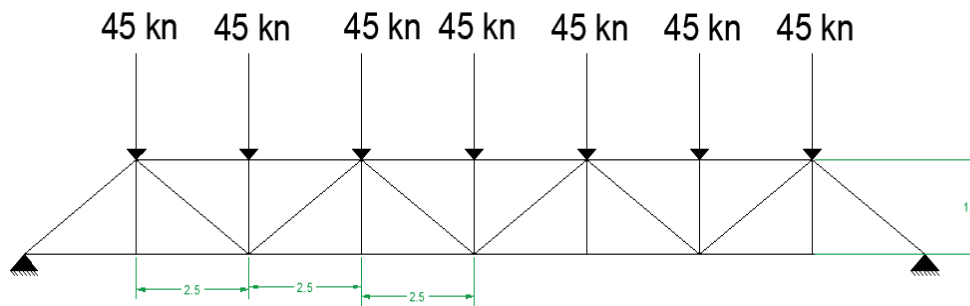


## 2 METHODOLOGY

Numerous crucial processes are involved in the design of steel trusses, such as comprehensive structural analysis, member selection, load analysis, and connection design. The following is the report's methodology:

1. Literature Review: Examining current studies and recommendations, such as those issued by the American Institute of Steel Construction (AISC), to identify best practices.
2. Load analysis: figuring out what kinds of loads (live, dead, wind, seismic) and how much of each the truss needs to support.
3. Preliminary Design: Creating preliminary member sizes and truss layouts in accordance with load calculations and architectural specifications.
4. Member and Connection Design: To guarantee strength and stability, the right steel sections should be chosen, and joints should be designed.
5. Analytical Methods: For initial force estimates, use techniques like the method of joints and method of sections.
- 6 Computational Analysis: Employing software tools (e.g., SAP2000, STAAD.Pro, ANSYS) for extensive modeling and analysis.
7. Optimization: Adapting the design to attain structural performance and material economy.
8. Case Studies: Examining particular instances of steel truss usage to highlight design concepts.

### 3 Case studies



truss from 8 spans each span 2.5 m length,  
height of truss is 1.8 m ,load

### Design of the truss

#### 1 first step analysis of the members

By analyses this truss we found this data for each member

member no	member length m	exile load in kn
1	2.5	-125
2	2.5	-125
3	2.5	125
4	2.5	125

5	2.5	125
6	2.5	125
7	2.5	-125
8	2.5	-125
9	3.08058	-269.551
10	2.5	-375
11	2.5	-375
12	2.5	-500
13	2.5	-500
14	2.5	-375
15	2.5	-375
16	3.08058	-269.551
17	1.8	0
18	3.08058	192.537
19	3.08058	-115.522
20	0	38.507
21	3.08058	38.507
22	0	-115.522
23	3.08058	192.537
24	0	-45
25	1.8	0
26	0	-45
27	0	0
28	1.8	-45
29	0	0

This date analyzed by sap 2000

Biggest member in compression is 500 kn

Biggest member in tension is 192.537 kn

Convert KN to kips

500Kn =112.4Kips

192.537 KN =43.28kips

## 2Design

### Design of compression member:

$P_u = 112.4 \text{ kips}$

The section we chose (Double Angle Section) with equal legs according to AISC specifications with  $(1/4'' \text{ thickness})$ . as shown in fig no 3

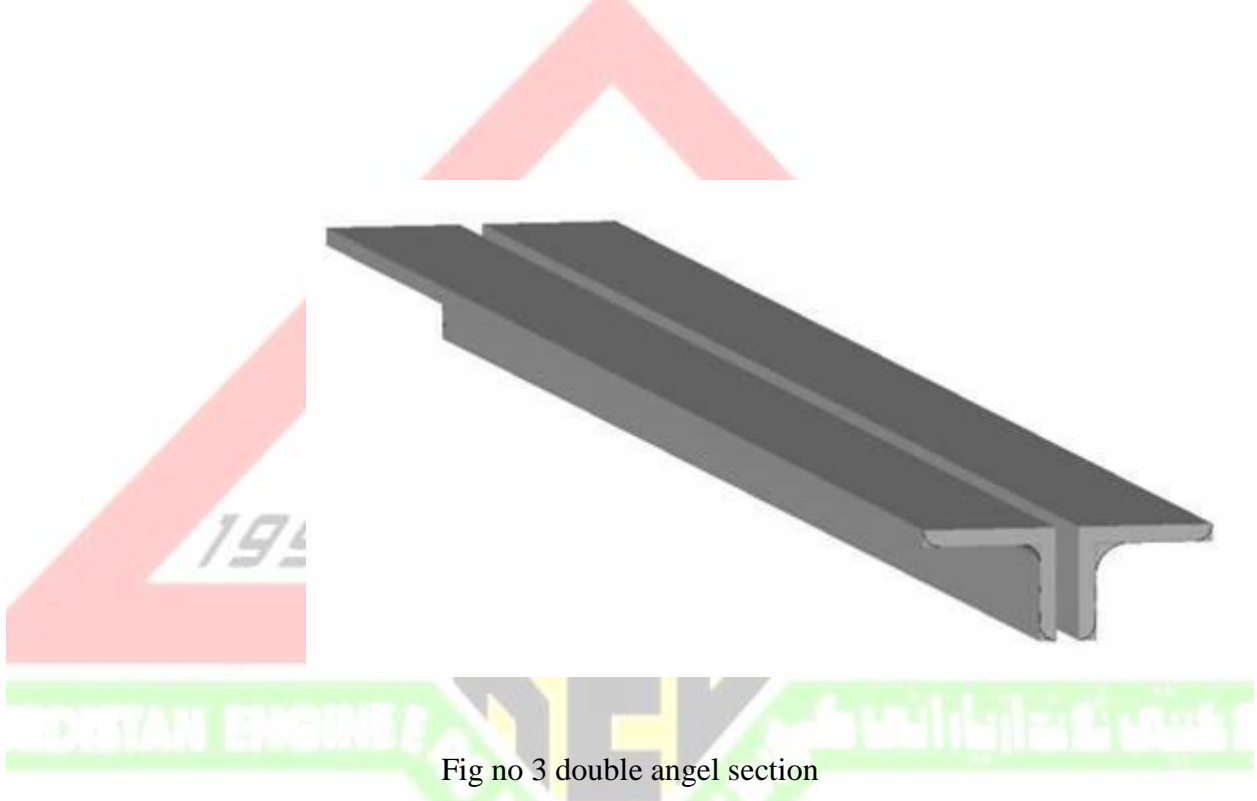


Fig no 3 double angel section

Because all joints are pin so  $(K) = 1$

Section  $2L 4 \times 4 \frac{1}{4}$

$$L = \frac{2.5}{0.3048} = 8.2 \text{ ft} \quad , K = 1$$

$$KL = 1 * 8.2 = 8.2$$

From table 4-8

$$\text{Section } 2L 4 \times 4 \frac{1}{4} \quad 2L 4 * 4 * \frac{1}{4} \rightarrow \phi P_n = 89.5 < 112.4 \therefore \text{not OK}$$



Section 2L 4x4 3/8  $\phi P_n = 134$

Check the section from table 4-22

From table 4-22  $\frac{KL}{r} = 80$ ,  $F_y = 36$

$$\phi_c F_{cr} = 22.7 \rightarrow \phi P_n = \phi_c F_{cr} * A_g$$

$$\phi P_n = 22.7 * 5.72 = 129.8 > 112.4 \text{ kips} \quad \therefore OK$$

$$\frac{KL}{r} = 80 < 4.71 \sqrt{\frac{E}{F_y}} \rightarrow 4.71 * \sqrt{\frac{29000}{36}} = 133.68; \frac{KL}{r} = 80 < 133.68$$

Ok

$$\therefore \text{use } F_{cr} = (0.658)^{\frac{F_y}{F_e}} * F_y; F_e = \frac{\pi^2 * E}{(KL/r)^2} = F_e = \frac{\pi^2 * 29000}{(80)^2} = 44.67$$

$$F_{cr} = (0.658)^{36/44.67} * 36 = 25.69$$

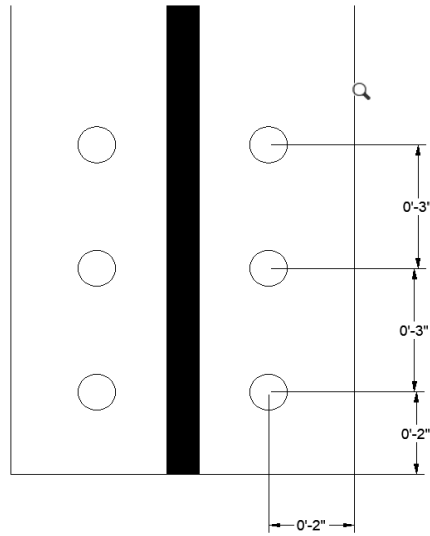
$$\phi_c P_n = \phi_c F_{cr} * A_g = 0.9 * 25.69 * 5.72 = 121.8 > 112.4 \quad \therefore OK$$

$$\frac{KL}{r} = 80 < 200 \quad \therefore OK$$

Check the local buckling: -

The shape (2L4\*4\*3/8) is not slender (there is not footnote in the dimensions and properties indicate that it is) so local buckling does not have to be investigated.

## Design of Tension members:



$$P_u = 192.537 \text{ KN} = 43.28 \text{ kips}$$

$$F_y = 36 \text{ kips}; \quad F_u = 58 \text{ kips}$$

Use double angle section

$$2L4 * 4 * \frac{3}{8}; \quad A_g = 5.72 \text{ in}^2$$

$$1) \phi P_n (\text{yield}) = 0.9 * F_y * A_g = 0.9 * 5.72 * 36 = 185.32 \text{ kips}$$

### Fracture Strength

$$A_n = A_g - 2 \left( \text{Area of one bolt} + \frac{1}{8} \right) * \text{Thickness}$$

$$A_n = 5.72 - \left( 2 \left( \frac{7}{8} + \frac{1}{8} \right) \right) * \frac{3}{8} = 4.97 \text{ inch}^2$$

From table 1.7

$$x' = 1.13, \quad u = 1 - \frac{1.13}{6} = 0.81$$

$$A_e = 4.97 * 0.81 = 4 \text{ in}^2$$

$$2) \phi P_n (\text{ultimate}) = 0.75 * F_u * A_e = 0.75 * 58 * 4 = 174 > 43 \therefore \text{OK}$$

**Check for block shear**

$$A_g(v) = 8 * \frac{3}{8} * 2 = 6in^2$$

$$A_n(v) = 6 - \left[ 2.5 \left( \frac{7}{8} + \frac{1}{8} \right) * \frac{3}{8} * 2 \right] = 4.125in^2$$

$$A_g(t) = 2 * \frac{3}{8} * 2 = 1.5in^2$$

$$A_n(t) = 1 - \left[ 0.5 * \frac{3}{8} * 2 \right] = 1.125in^2$$

$$1) \phi(0.6 * f_y * A_g(v) + F_u * A_n(t)) =$$

$$2) \phi(0.6 * F_u * A_n(v) + F_y * A_g(t)) =$$

$$3) \phi(0.6 * F_u * A_n(v) + F_y * A_g(t)) =$$

$$1) \quad 0.75(0.6 * 36 * 6 + 58 * 1.125) = 146.13kips$$

$$2) 0.75(0.6 * 58 * 4.125 + 58 * 1.125) = 156.6kips$$

$$4) \phi(0.6 * 58 * 4.125 + 36 * 1.125) = 138.1kips$$

$\therefore$  All of this results  $> P_u = 43kips \quad \therefore OK$

$$L/r = \frac{3.06/0.3048}{1.23} = 100 < 300 \quad \therefore OK$$

## Discussion and Result

From the result of the analysis and design of the members the section 2L4x4 3/8 it is suitable for all members and its best type according to the load on the member economical and safe

Below shown same data from analyzed truss by SAP200 that we used.

**Table: Active Degrees of Freedom**

**Table: Active Degrees of Freedom**

UX	UY	UZ	RX	RY	RZ
Yes	Yes	Yes	Yes	Yes	Yes

**Table: Analysis Options, Part 1 of 2**

**Table: Analysis Options, Part 1 of 2**

Solver	SolverProc	Force32Bit	StiffCase	GeomMod	HingeOpt	NumAThr eads	MaxFileSi ze	NumDThr eads
Advanced	Auto	No	None	None	In Elements	0	0	0

**Table: Analysis Options, Part 2 of 2**

**Table: Analysis Options, Part 2 of 2**

NumRThr eads	UseMMFil es	AllowDiff
0	Program Determined	No

**Table: Auto Wave 3 - Wave Characteristics - General**

**Table: Auto Wave 3 - Wave Characteristics - General**

WaveChar	WaveType	KinFactor	SWaterDepth	WaveHeight	WavePeriod	WaveTheory
			m	m	Sec	
Default	From Theory	1.	45.	18.	12.	Linear

**Table: Case - Modal 1 - General**

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Case	ModeType	MaxNum Modes	MinNum Modes	EigenShift	EigenCutoff	EigenTol	AutoShift
				Cyc/sec	Cyc/sec		
MODAL	Eigen	12	1	0.0000E+00	0.0000E+00	1.0000E-09	Yes
				0	0		

**Table: Case - Static 1 - Load Assignments**

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Case	LoadType	LoadName	LoadSF
POINT LOAD	Load pattern	POINT LOAD	1.

**Table: Connectivity - Frame, Part 1 of 2**

**Table: Connectivity - Frame, Part 1 of 2**

Frame	JointI	JointJ	IsCurved	Length	CentroidX	CentroidY	CentroidZ
				m	m	m	m
1	1	2	No	2.5	1.25	0.	0.
3	2	4	No	2.5	3.75	0.	0.
4	4	5	No	2.5	6.25	0.	0.

Table: Connectivity - Frame, Part 1 of 2

Frame	JointI	JointJ	IsCurved	Length	CentroidX	CentroidY	CentroidZ
				m	m	m	m
5	5	6	No	2.5	8.75	0.	0.
6	6	7	No	2.5	11.25	0.	0.
7	7	8	No	2.5	13.75	0.	0.
8	8	9	No	2.5	16.25	0.	0.
9	9	10	No	2.5	18.75	0.	0.
10	10	11	No	3.08058	18.75	0.	0.9
11	11	12	No	2.5	16.25	0.	1.8
12	12	13	No	2.5	13.75	0.	1.8
13	13	14	No	2.5	11.25	0.	1.8
14	14	15	No	2.5	8.75	0.	1.8
15	15	16	No	2.5	6.25	0.	1.8
16	16	17	No	2.5	3.75	0.	1.8
17	17	1	No	3.08058	1.25	0.	0.9
18	2	17	No	1.8	2.5	0.	0.9
19	17	4	No	3.08058	3.75	0.	0.9
20	4	15	No	3.08058	6.25	0.	0.9
21	15	6	No	3.08058	8.75	0.	0.9
22	6	13	No	3.08058	11.25	0.	0.9
23	13	8	No	3.08058	13.75	0.	0.9
24	8	11	No	3.08058	16.25	0.	0.9
25	4	16	No	1.8	5.	0.	0.9
26	5	15	No	1.8	7.5	0.	0.9
27	6	14	No	1.8	10.	0.	0.9
28	7	13	No	1.8	12.5	0.	0.9
29	8	12	No	1.8	15.	0.	0.9
30	9	11	No	1.8	17.5	0.	0.9



**Table: Frame Output Station Assignments**

**Table: Frame Output Station Assignments**

Frame	StationType	MinNumSta	MaxStaSp	AddAtElm	AddAtPtL
	g		cg	Int	oad
			m		
1	MaxStaSp		0.5	Yes	Yes
	g				
3	MaxStaSp		0.5	Yes	Yes
	g				
4	MaxStaSp		0.5	Yes	Yes
	g				
5	MaxStaSp		0.5	Yes	Yes
	g				
6	MaxStaSp		0.5	Yes	Yes
	g				
7	MaxStaSp		0.5	Yes	Yes
	g				
8	MaxStaSp		0.5	Yes	Yes
	g				
9	MaxStaSp		0.5	Yes	Yes
	g				
10	MinNumSta	3		Yes	Yes
	a				
11	MaxStaSp		0.5	Yes	Yes
	g				
12	MaxStaSp		0.5	Yes	Yes
	g				
13	MaxStaSp		0.5	Yes	Yes
	g				
14	MaxStaSp		0.5	Yes	Yes
	g				
15	MaxStaSp		0.5	Yes	Yes
	g				
16	MaxStaSp		0.5	Yes	Yes
	g				

**Table: Frame Output Station Assignments**

Frame	StationType	MinNumSta	MaxStaSpacing	AddAtElmInt	AddAtPtLoad
17	MinNumSta	3		Yes	Yes
18	MinNumSta	3		Yes	Yes
19	MinNumSta	3		Yes	Yes
20	MinNumSta	3		Yes	Yes
21	MinNumSta	3		Yes	Yes
22	MinNumSta	3		Yes	Yes
23	MinNumSta	3		Yes	Yes
24	MinNumSta	3		Yes	Yes
25	MinNumSta	3		Yes	Yes
26	MinNumSta	3		Yes	Yes
RAMPTH		4.	1.		
UNIFTH		0.	1.		
UNIFTH		1.	1.		

Table: Grid Lines, Part 1 of 2

Table: Grid Lines, Part 1 of 2							
CoordSys	AxisDir	GridID	XRYZCoord	LineType	LineColor	Visible	BubbleLoc
GLOBAL	X	A	0.	Primary	Gray8Dark	Yes	End
GLOBAL	X	B	2.5	Primary	Gray8Dark	Yes	End
GLOBAL	X	C	5.	Primary	Gray8Dark	Yes	End
GLOBAL	X	D	7.5	Primary	Gray8Dark	Yes	End
GLOBAL	X	E	10.	Primary	Gray8Dark	Yes	End
GLOBAL	X	F	12.5	Primary	Gray8Dark	Yes	End
GLOBAL	X	G	15.	Primary	Gray8Dark	Yes	End
GLOBAL	X	H	17.5	Primary	Gray8Dark	Yes	End
GLOBAL	X	I	20.	Primary	Gray8Dark	Yes	End
GLOBAL	Y	1	0.	Primary	Gray8Dark	Yes	Start
GLOBAL	Z	Z1	0.	Primary	Gray8Dark	Yes	End
GLOBAL	Z	Z2	1.8	Primary	Gray8Dark	Yes	End

Table: Grid Lines, Part 2 of 2

Table: Grid Lines, Part 2 of 2		
CoordSys	AllVisible	BubbleSize
GLOBAL	Yes	0.4375
GLOBAL		
GLOBAL		
GLOBAL		
GLOBAL		
GLOBAL		
GLOBAL		
GLOBAL		

**Table: Grid Lines, Part 2 of 2**

CoordSys	AllVisible	BubbleSize
GLOBAL		e
GLOBAL		m
GLOBAL		

**Table: Groups 1 - Definitions, Part 1 of 3**

**Table: Groups 1 - Definitions, Part 1 of 3**

GroupName	Selection	SectionCut	Steel	Concrete	Aluminum	ColdFormed	Staged
ALL	Yes	Yes	Yes	Yes	Yes	Yes	Yes

**Table: Groups 1 - Definitions, Part 2 of 3**

**Table: Groups 1 - Definitions, Part 2 of 3**

GroupName	Bridge	AutoSeismic	AutoWind	SelDesSteel	SelDesAluminum	SelDesCold	MassWeight
ALL	Yes	No	No	No	No	No	Yes

**Table: Groups 1 - Definitions, Part 3 of 3**

**Table: Groups 1 - Definitions,  
Part 3 of 3**

GroupName	Color
ALL	Red

**Table: Groups 3 - Masses and Weights**

**Table: Groups 3 - Masses and Weights**

GroupName	SelfMass	SelfWeight	TotalMass	TotalMass	TotalMass
			X	Y	Z
	KN-s2/m	KN	KN-s2/m	KN-s2/m	KN-s2/m
ALL	0.87	8.521	0.87	0.87	0.87

**Table: Joint Coordinates, Part 1 of 2**

**Table: Joint Coordinates, Part 1 of 2**

Joint	CoordSys	CoordType	XorR	Y	Z	SpecialJt	GlobalX
			m	m	m		m
1	GLOBAL	Cartesian	0.	0.	0.	No	0.
2	GLOBAL	Cartesian	2.5	0.	0.	No	2.5
4	GLOBAL	Cartesian	5.	0.	0.	No	5.
5	GLOBAL	Cartesian	7.5	0.	0.	No	7.5
6	GLOBAL	Cartesian	10.	0.	0.	No	10.
7	GLOBAL	Cartesian	12.5	0.	0.	No	12.5
8	GLOBAL	Cartesian	15.	0.	0.	No	15.
9	GLOBAL	Cartesian	17.5	0.	0.	No	17.5
10	GLOBAL	Cartesian	20.	0.	0.	No	20.
11	GLOBAL	Cartesian	17.5	0.	1.8	No	17.5
12	GLOBAL	Cartesian	15.	0.	1.8	No	15.
13	GLOBAL	Cartesian	12.5	0.	1.8	No	12.5
14	GLOBAL	Cartesian	10.	0.	1.8	No	10.
15	GLOBAL	Cartesian	7.5	0.	1.8	No	7.5
16	GLOBAL	Cartesian	5.	0.	1.8	No	5.
17	GLOBAL	Cartesian	2.5	0.	1.8	No	2.5

**Table: Joint Coordinates, Part 2 of 2**

**Table: Joint Coordinates, Part 2 of 2**

Joint	GlobalY m	GlobalZ m	GUID
1	0.	0.	77f10cb4-75c1-4d4c- b7b5-d831129fd35d
2	0.	0.	6d9752b7-86b8-45ad- 88d7-bda2e6fdc196

## Conclusion

For this type of truss that we designed the compression member controlled the truss because the load on the compression member larger than the tension so chosen the section 2L4x4 3/8 that controlled all members

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